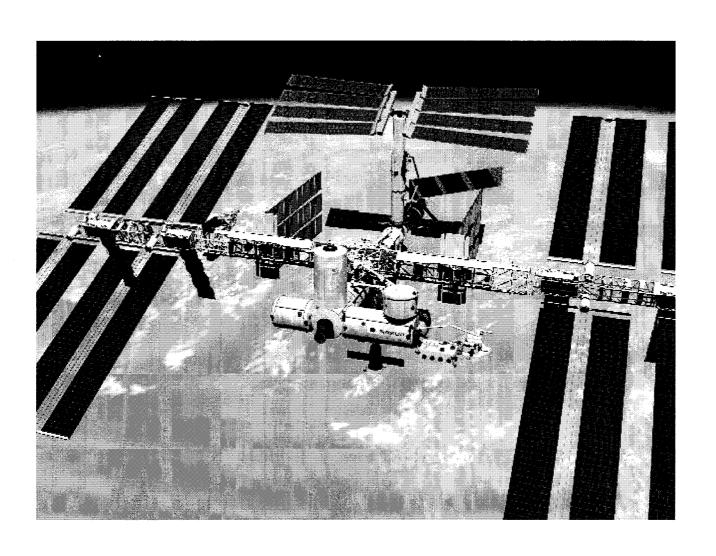
# Clock Technology Development in the Laser Cooling and Atomic Physics (LCAP) Program

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## **Credits**

**JPL:** LCAP program

Dave Seidel (Systems Engineer)
Rob Thompson (Instrument Manager)
Lute Maleki (Group Sup./Proj. Scientist)
Jim Kohel
Bill Klipstein

GPS Carrier Phase: Larry Young Sien Wu

Project Management:
Mike Devirian (Program Manager)
Gail Klein (Project Manager)
Ed Dobkowski (Quality Assurance)
Ulf Israelsson (Discipline Scientist)
Richard Beatty (ISS Program Engineer)

Yale: GLACE, RACE, LCATS

**Kurt Gibble** 

**Don Sullivan** 

**NIST:** PARCS, LCATS

Tom Heavner

Leo Hollberg

Steve Jefferts

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Dawn Meekhof

Craig Nelson

Tom Parker

William Phillips

**Hugh Robinson** 

Steve Rolston

Fred Walls

Andrea De Marchi (Torrino)

CU: PARCS, LCATS

**Neil Ashby** 

**SAO: PARCS** 

Bob Vessot Ed Mattison



# Overview of LCAP Flight Projects

#### **International Space Station**

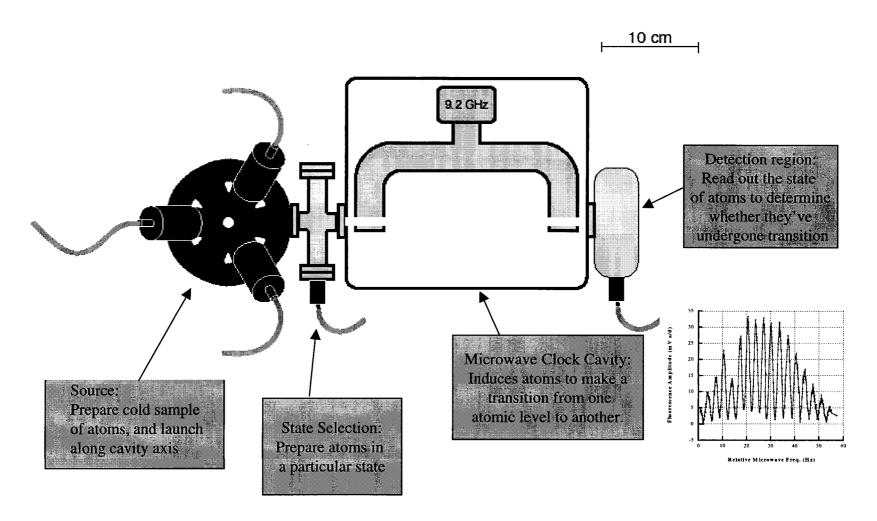
- •PARCS (Primary Atomic Reference Clock in Space): NIST/CU
  Laser-cooled cesium primary frequency standard (10<sup>-16</sup> accuracy) operating continuously for at least 30 days, with GPS capability. Will perform relativity experiments and global precise time distribution.
- •RACE (Rubidium Atomic Clock Experiment): Yale
  Laser-cooled rubidium clock for ultrahigh accuracy (exceeding a part in 10<sup>16</sup>), to operate continuously for at least 30 days. Use of clock for relativity experiments and cold collision studies.

#### Space Shuttle

•LCATS (Laser Cooled Atomic Timekeeping in Space): Joint PARCS/RACE team. Flight of laser-cooled microgravity atomic clock along with high stability ion clock/H maser and GPS capability for relativity experiments, tests of spatial isotropy. Tests time transfer and clock technology with some science return.



# Space Clock 101



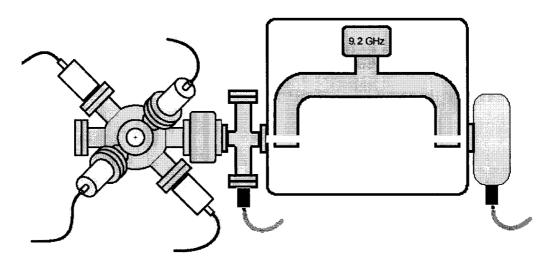


# Physics with Clocks in microgravity

- •Gravitational frequency shift (requires stable frequency transfer to ground)
- •Local Position Invariance (requires comparison to another oscillator)
- •Kennedy-Thorndike Experiment (requires cavity oscillator such as SUMO)



# Space Clock Challenges



### Laser Cooling Source

Lasers

**Optical Frequency Control** 

**Fibers** 

Fluorescence detection

Vacuum chamber

Computer Control

Electronics

Magnetic field control

**Atom Source** 

#### **Clock Parts**

Microwave electronics

Local Oscillator

Synthesizer

Cavity

More magnetic field control

Thermal Control

Light Baffling/Shutters

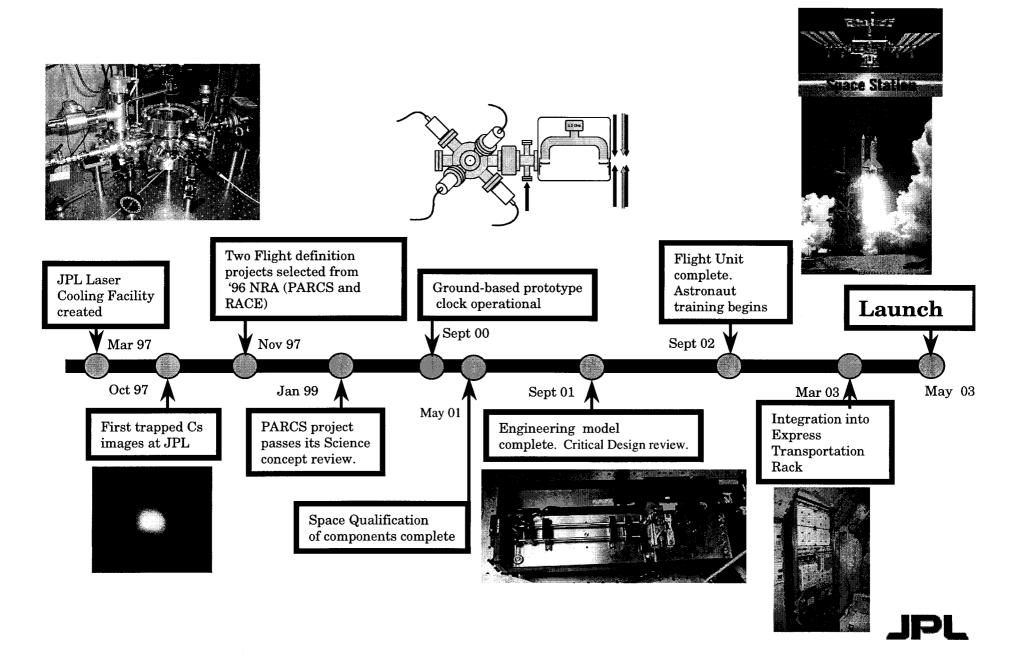
Vacuum requirements

Measurement System

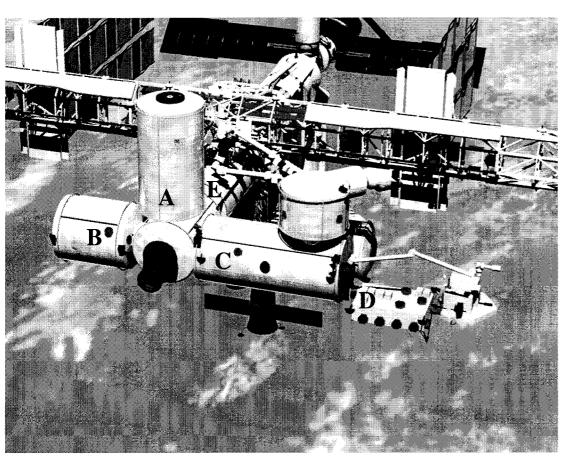


#### JPL

## **LCAP** Timeline



## **ISS Science Platforms**



- **A) Centrifuge Accommodation Module**
- **B)**Columbus Orbiting Facility
- C)Japanese Experiment Module (JEM)
- D) JEM external facility
- E) US Lab

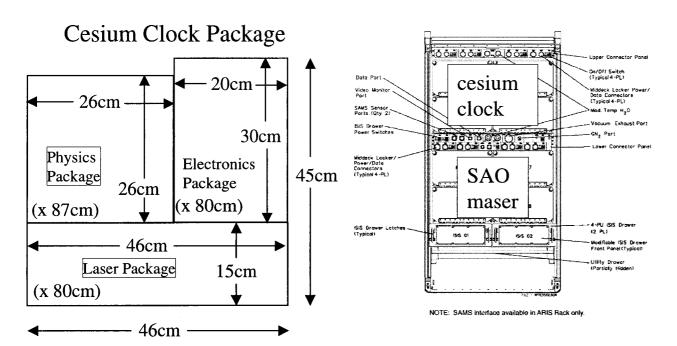
Not shown: Russian Laboratories, Express Pallets

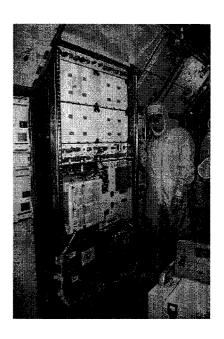


## **PARCS**

# ISS Express Rack







	Requirement	Constraint	Reserve
Mass	130Kg	195Kg	65Kg
Power	< 500W	<2kW	1.5kW
Volume	162 liters	248 liters	86 liters
length	87cm	90.7cm	3.7cm
depth	46cm	51.6cm	5.6cm
height	45cm	53.1cm	8.1cm



# Space Qualification of Components

### Shuttle requirements:

#### •Vibration Testing:

Instrument should operate after exposure to:

Freq. Range Design/Protoflight (PF) Flight Acceptance (FA)

20 to 150 Hz +6dB/Octave +6dB/Octave 150 to 1000 Hz 0.06 g<sup>2</sup>/Hz 0.03 g<sup>2</sup>/Hz 1000 to 2000 Hz -6dB/Octave -6dB/Octave

Duration: Design: 2 minutes; PF or FA test: 1 minute

#### •Environment:

Instrument should operate after exposure to:

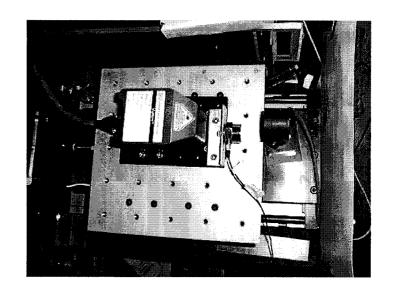
Temperature: -5 to 50 C

Pressure: 786 torr to 204 torr (1240 torr/min Max Depressurization rate)

Humidity: 20 to 70%

#### •Radiation:

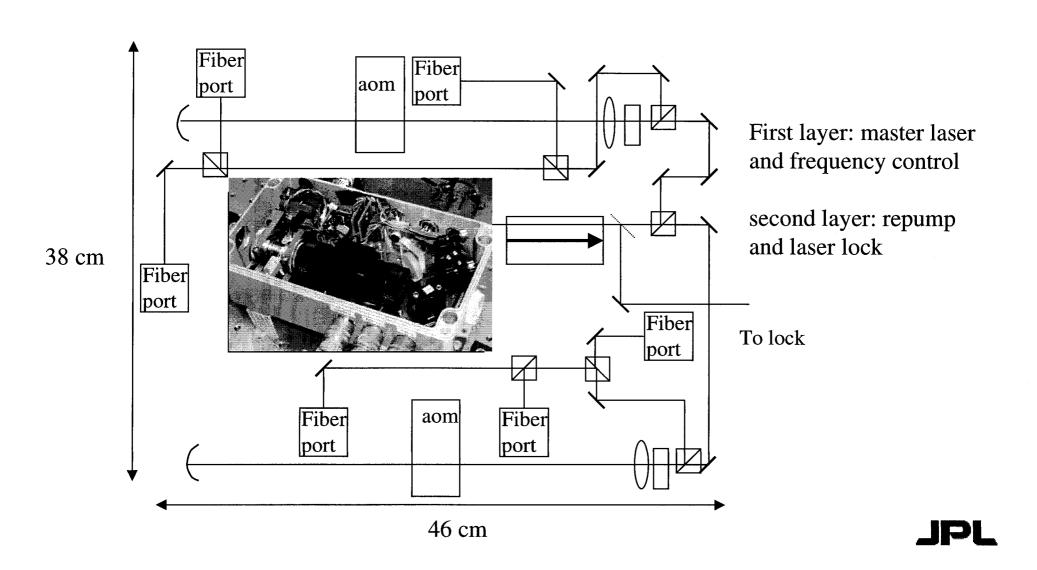
~100x Earth dose



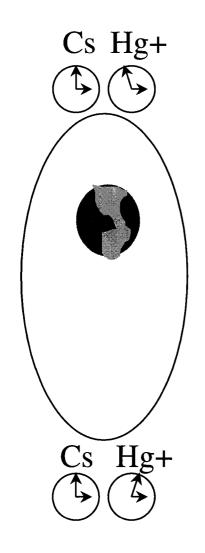
New Focus Vortex laser on vibration test bed at JPL

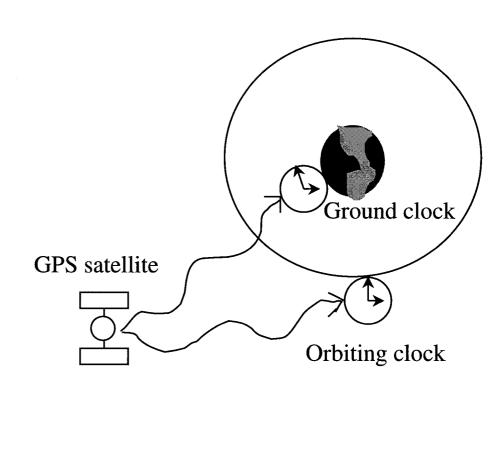


# Laser Configuration



# Clock Rate Comparisons: GPS Carrier Phase Frequency Transfer







# GPS Carrier Phase Frequency Transfer

GPS carrier phase technique expected to give:

- •100 ps resolution
- •< 10 cm position information
- •< 1mm/s velocity information

#### **Issues:**

- •Need external antennae
- •No high quality rf/optical link between interior/exterior
- •Multipath worrisome (need ~-70 dBm)
- •visibility of satellites (desire ~12 in view)

Existing GPS antennae will see between 3-6 satellites Give Position Information to 100 m



## ISS Model Views

"Normal" View

Another "Normal" View

